



# **SERVICE BULLETIN**

SB No. L-410/041b Aeroplane Type: L-410

Revision No. IR Date of Issue: Oct 8, 2020

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Classification: Recommended

**Subject**: Installation of a ball valve into return pipe from an engine drainage sump.

Supersedure: N/A

### 1. PLANNING INFORMATION

### A. APPLICABILITY

L 410 UVP-E20 aeroplanes up to S/N 3220 inclusive.

### **B. CONCURRENT REQUIREMENTS**

None.

### C. REASON

To prevent fuel leakage from the fulfilled fuel tanks when the ambient temperature rises.

# D. COMPLIANCE TIME

As per operator's requirement.

### E. MANPOWER

Supposed Man-hours: 3 M-hours

### F. MATERIAL

Will be delivered by the aeroplane manufacturer Aircraft Industries, a.s., 686 04 Kunovice, Czech Republic against an order.

# **G. NEW EQUIPMENT**

None.



### **H. SPECIAL TOOLS**

None.

### I. EXPENSES

To be covered by the operator.

# J. WEIGHT AND BALANCE

Influence of the installation on the empty aeroplane weight and balance is negligible.

# **K. REFERENCE DOCUMENTS**

# 1. Drawings

B562 500LP sheet 1 Fuel system installation in the wing

B562 527LP Lid

B562 658LP sheet 1, 7 Fuel system installation in the wing

B564 527LP Lid assembled

# 2. Maintenance Manual, Doc. No. Do-L410-1232.2, chap.:

# L. AFFECTED DOCUMENTS

AFM not affected MS not affected MM not affected WM not affected

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# 2. IMPLEMENTATION INFORMATION

#### A. PREPARATORY WORK

- Disconnect the external power supply plug according to the MM, chap. 024.40.00.B.
- Remove accumulators from the airplane according to the MM, chap. 024.30.04.A.
- Drain fuel from the fuel system.
- Remove access lids from upper wing skin between ribs 6 and 7.

#### **B. CHECK OR INSPECTION**

None.

#### C. DISASSEMBLY

- Loosen screws and disconnect the venting hose and fuel return hose from the middle fuel tank. Blind the holes.
- Unscrew nuts and screws and remove lids pos. 28, Dwg No. B562 500LP sheet 1. Keep connecting material for reassembly.
- Remove pos. 2, 5, 7, Dwg No. B562 527LP from the lids and enlarge the original hole
  Ø 10 mm to Ø 30 mm. Deburr the holes and treat them by Bonderite M-CR 1200S Aero
  according to the MM, chap. 020.12.00 and coat by primer acc. to the chap. 020.20.00.
  Use material pos. 1 or 2 of point 3.B. of this bulletin.

## D. ASSEMBLY

- Install the elbow pipe pos. 3, sealing ring pos. 6, screw pos. 4 and ball pos. 8 on the original lids according to the Dwg No. B564 527LP, Detail I. Secure the screw by Loctite 243 and fix the ball by a split pin pos. 9. Use material pos. 1 to 5 of point 3.A. of this bulletin and pos. 3 of point 3.B.
- Install back the lids pos. 28, Dwg No. B562 658LP sheet 1. Make conductive connection according to the View U7, sheet 7. Use original connecting material.
- Remove blinds and connect the venting hose and fuel return hose to the middle fuel tank. Carefully tighten the screws of hose sleeves.

#### E. TESTS

#### 1. Ground Tests

None.

## 2. Flight Tests

None.

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# F. COMPLETION WORK

- Repair damaged paintings with paint according to the MM, chap. 020.20.00.
- Install access lids on upper wing skin between ribs 6 and 7.
- Install the airplane accumulators according to the MM, chap. 024.30.04.A.

# 3. BILL OF MATERIAL

# A. MATERIAL DELIVERED BY THE AEROPLANE MANUFACTURER

Pos.	Nomenclature	Name	Pcs/ArpIn
1	B564525N	Elbow-pipe	2
2	B660229N	Screw	2
3	B660230N	Ball	2
4	81119086	30x3 O-ring, ONL 3973.2	2
5	36106486	2.5x36 Split pin, DIN 94	2

#### **B. MATERIAL SUBJECT TO SPECIAL CONDITIONS DURING TRANSPORT**

Pos.	Nomenclature	Name	Qty/ArpIn
1	82055550	Bonderite M-CR 1200S Aero or	50 g
2	82055560	Bonderite M-CR 1132 Pen	1
3	82177508	Loctite 243	5 g

# 4. RECORD IN THE AEROPLANE LOGBOOK

Record implementation of this L-410/041b Service Bulletin.

Date:	Accomplished by:
	(name and signature of authorised staff)

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